**Control Laboratory:** 

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## Lezione $15-13~\mathrm{April}$

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### 15.1 Internal Model Principle (continued)

In the last lecture we have began to study the internal model principle, which is a generalization of the integral control. Our purpose is to track a certain signal  $(y(t) \rightarrow r(t))$  and reject another signal (w(t)), which satisfy the following property:

$$r^{(m)} + \alpha_{m-1}r^{(m-1)} + \dots + \alpha_0 r = 0$$
(15.1)

$$w^{(m)} + \alpha_{m-1}w^{(m-1)} + \dots + \alpha_0 w = 0$$
(15.2)

,

,

where  $\{\alpha_i\}_{i=0}^{m-1}$  are known.

We have introduced a new state: 
$$z \stackrel{\triangle}{=} \begin{bmatrix} e \\ \vdots \\ e^{(m-1)} \\ \hline \xi \end{bmatrix} \in \mathbb{R}^{(n+m)}$$
  
ere

where

$$e^{(k)} = Cx^{(k)} - r^{(k)}$$
,  
 $\xi \stackrel{\triangle}{=} x^{(m)} + \alpha_{m-1}x^{(m-1)} + \dots + \alpha_0 x$ ,

because with this new representation the reference signal and the input disturbance vector disappear. We found:

$$\dot{z} = \begin{bmatrix} e^{(1)} \\ \vdots \\ e^{(m)} \\ \hline \xi \end{bmatrix} = \overbrace{\left[ \begin{array}{ccccc} 0 & 1 & 0 & \cdots & 0 & 0 \\ 0 & 0 & 1 & \cdots & 0 & 0 \\ \vdots & \vdots & \vdots & \ddots & \vdots & \vdots \\ 0 & 0 & 0 & \cdots & 1 & 0 \\ \hline -\alpha_0 & -\alpha_1 & -\alpha_2 & \cdots & -\alpha_{m-1} & C \\ \hline 0 & 0 & 0 & \cdots & 0 & | A \end{bmatrix}} \overbrace{\left[ \begin{array}{c} e \\ \vdots \\ e^{(m-1)} \\ \hline \xi \end{bmatrix}}^{B_z} + \overbrace{\left[ \begin{array}{c} 0 \\ 0 \\ \vdots \\ 0 \\ \hline 0 \\ \hline B \end{bmatrix}}^{u_{\xi}} \quad ,$$

with

$$u_{\xi} \stackrel{\triangle}{=} u^{(m)} + \alpha_{m-1}u^{(m-1)} + \dots + \alpha_0 u,$$

so  $\dot{z} = A_z z + B_z u_{\xi}$ .

If we introduce a negative state feedback we obtain:  $\left\{ \begin{array}{l} \dot{z}=A_zz+B_zu_\xi\\ u_\xi=-K_zz \end{array} \right..$ 

If  $(A_z, B_z)$  is reachable, then  $\exists K_z \in \mathbb{R}^{(m+n) \times p} : (A_z - B_z K_z)$  is asymptotically stable, because it has eigenvalues that can be arbitrarily placed. So  $z(t) \to 0$  for each initial condition of r and w that satisfy 15.1 e 15.2. Therefore  $e(t) \to 0$ , that is  $y(t) \to r(t)$ .

If we apply the PBH test for the reachability of  $(A_z, B_z)$  we find:

ſ	s	-1	0	•••	0	0	0	
	0	s	-1	• • •	0	0	0	
	÷	÷	÷	·	÷	:	:	
l	0	0	0	• • •	-1	0	0	,
	$\alpha_0$	$\alpha_1$	$\alpha_2$	•••	$s + \alpha_{m-1}$	-C	0	
	0	0	0		0	sI - A	B	

that must have a rank equal to n+m.

The necessary and sufficient conditions which guarantee that  $(A_z, B_z)$  is reachable are:

- (A, B) reachable;
- zeroes of  $(s^m + \alpha_{m-1}s^{m-1} + ... + \alpha_0)$  are not zeroes of the transfer function of the original system (A, B, C).

Indeed (A, B) is reachable, the first m-1 rows and the last n rows of the PBH matrix are always linearly independent for each  $s \in \mathbb{C}$ . But we have to verify when the m-th row is linearly independent from all others. If  $s = \overline{z}$  is a zero of  $s^m + \ldots \alpha_0 = 0$ , then the first m components of the m-th row are linearly dependent from the first m-1rows. Besides if  $s = \overline{z}$  is also a zero of the transfer function of the system (A, B, C), then the last n components of the m-th row are linearly dependent from the last nrows of the PBH matrix, and so the rank of the latter decreases to m + n - 1.

Now we want to describe the system with this new type of control as a function of x(t) and of the error signal e(t). Considering that:

$$u_{\xi} = -\underbrace{\left[\begin{array}{ccc} k_0 & \cdots & k_{m-1} \\ & & \\ \end{array}\right]}_{\in \Re^m} \underbrace{\left[\begin{array}{c} e \\ \vdots \\ e^{(m-1)} \\ \xi \end{array}\right]}_{\xi}$$

If we expand  $u_{\xi}$ :

$$u^{(m)} + \alpha_{m-1}u^{(m-1)} + \dots + \alpha_0 u = -k_0 e - \dots - k_{m-1}e^{(m-1)} - k_{\xi}(x^{(m)} + \alpha_{m-1}x^{(m-1)} + \dots + \alpha_0 x)$$

We can collect some terms in this way:

$$(u^{(m)} + k_{\xi}x^{(m)}) + \alpha_{m-1}(u^{(m-1)} + k_{\xi}x^{(m-1)}) + \dots + \alpha_0\underbrace{(u+k_{\xi}x)}_{\tilde{u}} = -k_0e - \dots - k_{m-1}e^{(m-1)}$$

So we obtain:

$$\widetilde{u}^{(m)} + \alpha_{m-1}\widetilde{u}^{(m-1)} + \dots + \alpha_0\widetilde{u} = -k_0e - \dots - k_{m-1}e^{(m-1)}$$

Applying the Laplace Transform:

$$s^{m}\widetilde{U}(s) + \alpha_{m-1}s^{m-1}\widetilde{U}(s) + \dots + \alpha_{0}\widetilde{U}(s) = -k_{0}E(s) - \dots - k_{m-1}s^{m-1}E(s)$$

So

$$\widetilde{U}(s) = -\underbrace{\frac{k_{m-1}s^{m-1} + \dots + k_0}{s^m + \alpha_{m-1}s^{m-1} + \dots + \alpha_0}}_{P_e(s)} E(s)$$
$$= -P_e(s)E(s)$$

The complete scheme of the overall system controlled by using the internal control principle is reported in Fig.15.1.



**Figura 15.1.** Scheme of the internal model control. (Typo: the error e(t) should be replaced with -e(t) in the figure)

We can notice that this is a generalization of the integral control in which we have assumed that the state x(t) is accessible. Indeed in the case of integral control we have:

$$P_e(s) = \frac{k_I}{s} = \frac{k_o}{s}.$$

#### 15.1.1 Summarizing

- 1. Check if (A, B) is reachable.
- 2. Check that roots of  $(s^m + \alpha_{m-1}s^{m-1} + \dots + \alpha_0)$  are not zeroes of  $P(s) = C(sI A)^{-1}B$ .
- 3. Build matrices  ${\cal A}_z$  ,  ${\cal B}_z.$
- 4. Decide where to place eigenvalues: using dominant pole approximation with performance region.
- 5.  $K_z = [k_0 \dots k_{m-1} | k_z]$ , so compute  $P_e(s)$ .

Considering the transfer function of the signal of the closed loop system:

$$Y(s) = P_{ry}(s)R(s)$$

If we are assume to track a sinusoidal signal  $r(t) = a \sin(\omega_0 t + \phi)$ , at steady state we will have  $y(t) = a |P_{ry}(j\omega_0)| \sin(t + \phi + \angle P_{ry}(j\omega_0))$ .

Instead if we have the disturbance signal  $w(t) = a \sin(t\omega_0 + \phi)$  and r(t) = 0, the output will go to zero  $(y(t) \to 0)$ . So the transfer function  $Y(s) = P_{wy}(s)W(s)$  must have a zero in  $\omega_0$ :  $P_{wy}(j\omega_0) = 0$ .



**Figura 15.2.** Module of the transfer function  $P_{ry}(j\omega)$ .



Figura 15.3. Phase of the transfer function  $P_{ry}(j\omega)$ .



Figura 15.4. Module of the transfer function  $P_{wy}(j\omega)$ 

# 15.2 Observers and Regulators

$$\begin{cases} \dot{x} = Ax + Bu\\ y = Cx \end{cases}$$
$$\int \dot{x} = A\hat{x} + Bu$$

$$x = Ax + Bi$$
$$y = C\hat{x}$$

which are the state rappresentation of Plant and Observer.



Figura 15.5. Scheme of Plant and observer.

State error:  $e_x = x - \hat{x}$ measuring error:  $e_y = y - \hat{y} = C(x - \hat{x}) = Ce_x$  $\dot{e_x} = Ae_x \Rightarrow \begin{array}{c} e \to 0 \\ \hat{x} \to x \end{array} \Leftrightarrow A \text{ is stable}$  $\dot{e_x} = \dot{x} - \dot{\hat{x}} = Ax + Bu - (A\hat{x} + Bu + Le_y) = (A - LC)e_x$ If (A - LC) is strictly stable  $\Rightarrow \hat{x}(t) \to x(t) \ \forall u(t), \forall A$ 



 $\Rightarrow$  we can use the state feedback control.



Figura 15.6. Scheme of Plant, observer and controller.

$$\begin{cases} \dot{x} = Ax + Bu \\ y = Cx \\ \dot{\hat{x}} = A\hat{x} + Bu + L(y - \hat{y}) \\ \hat{y} = C\hat{x} \\ u = u_{in} + u_c \\ u_c = -k\hat{x} \end{cases}$$
 equations of the system

$$\underbrace{\begin{array}{c} \mathbf{U}_{\mathrm{in}} \\ \mathbf{Y} = \mathbf{C}_{z} \mathbf{Z} \end{array}}_{\mathbf{Y} = \mathbf{C}_{z} \mathbf{Z}} \underbrace{\begin{array}{c} \mathbf{Y} \\ \mathbf{Y} = \mathbf{C}_{z} \mathbf{Z} \end{array}}_{\mathbf{Y} = \mathbf{C}_{z} \mathbf{Z}} \underbrace{\begin{array}{c} \mathbf{Y} \\ \mathbf{Z} \end{array}}_{\mathbf{Y} = \mathbf{C}_{z} \mathbf{Z}} \underbrace{\begin{array}{c} \mathbf{Z} \\ \mathbf{Z} \end{array}}_{\mathbf{Z} = \mathbf{Z}} \underbrace{\begin{array}{c} \mathbf{Z} \\ \mathbf{Z} \end{array}}_{\mathbf{Z} \mathbf{Z}} \underbrace{\begin{array}{c} \mathbf{Z} \\ \mathbf{Z} \end{array}}_{\mathbf{Z}} \underbrace{\begin{array}{c} \mathbf{Z} \\ \mathbf{Z} \end{array}}_{\mathbf{Z} \mathbf{Z}} \underbrace{\begin{array}{c} \mathbf{Z} \end{array}}_{\mathbf{Z} \mathbf{Z}} \underbrace{\begin{array}{c} \mathbf{Z} \\ \mathbf{Z} \end{array}}_{\mathbf{Z}} \underbrace{\begin{array}{c} \mathbf{Z} \end{array}}_{\mathbf{Z} \mathbf{Z}} \underbrace{\begin{array}{c} \mathbf{Z} \end{array}}_{\mathbf{Z}} \underbrace{\begin{array}{c} \mathbf{Z} \end{array}}_{\mathbf{Z} \mathbf{Z}} \underbrace{\begin{array}{c} \mathbf{Z} \end{array}}_{\mathbf{Z} \mathbf{Z}} \underbrace{\begin{array}{c} \mathbf{Z} \end{array}}_{\mathbf{Z} \mathbf{Z}} \underbrace{\begin{array}{c} \mathbf{Z} \end{array}}_{\mathbf{Z}} \underbrace{\end{array}}_{\mathbf{Z}} \underbrace{\begin{array}{c} \mathbf{Z} \end{array}}_{\mathbf{Z}} \underbrace{\end{array}}_{\mathbf{Z}} \underbrace{\end{array}}_{\mathbf{Z}} \underbrace{\begin{array}{c} \mathbf{Z} \end{array}}_{\mathbf{Z}} \underbrace{\end{array}}_{\mathbf{Z}} \underbrace{\end{array}}_{\mathbf{Z}$$

from the equations we obtain:  $\dot{x} = Ax + Bu_{in} - Bk\hat{x} = Ax + Bu_{in} - Bk(x - e_x)$   $\dot{e_x} = Ax + Bu_{in} - Bk\hat{x} - (A\hat{x} + Bu_{in} - Bk\hat{x} + LCe_x) = (A - LC)e_x$ 

### 15.2.1 Dynamical system

The state space representation with the more useful state z is:

$$\dot{z} = \begin{bmatrix} \dot{x} \\ \dot{e_x} \end{bmatrix} = \begin{bmatrix} A - BK & BK \\ 0 & A - LC \end{bmatrix} \begin{bmatrix} x \\ e_x \end{bmatrix} + \begin{bmatrix} B \\ 0 \end{bmatrix} u_{in}$$
$$y = \begin{bmatrix} C & 0 \end{bmatrix} \begin{bmatrix} x \\ e_x \end{bmatrix}$$

The stability of these system depends on the eigenvalues of  $A_z$ :

$$\Re[\lambda(A_z)] < 0 \Leftrightarrow Stable$$

The matrix is upper triangular so we can, by using the separation principle, divide the eigenvalues of  $A_z$  in the union between the eigenvalues of the controller and the eigenvalues of the observer:

$$\lambda(A_z) = \lambda(A - BK) \bigcup \lambda(A - LC)$$

 $\begin{cases} (A,B), REACHABLE \Rightarrow \\ (A,C), OBSERVABLE \Rightarrow \end{cases}$  Exists K, L such that eigenvalues can be placed arbitrarily.

We will choose, in order to stabilize the system, eigenvalues with  $Re[\lambda(A_z)] < 0$ .

#### 15.2.2 Tracking of a reference signal

How can be the matrices L and K chosen in order to achieve the best performance possible? How can be  $\lambda(A - LC)$  chosen?

In order that  $\hat{x} \to x$  as fast as possible,  $\lambda(A - LC)$  has to be chosen more negative (faster) than  $\lambda(A - BK)$ .

It is important to notice that L cannot be chosen too big because it amplifies the measure noise and because in that case the error  $e_x$  can have high peaks in the initial phase of the transitory. This peaks can force the controller to have high control signals.

$$\dot{x}(t) = (A - LC)x(t) + Bu(t) + L(y(t) + d_y(t))$$

with  $d_{u}(t)$  representing the measurement noise.

A rule of thumb for the design is to choose the eigenvalues of the observer matrix 3-10 times faster than the ones of the controller, in order to have a good compromise between the velocity of the tracking and the rejection of the output disturbances:

$$|\lambda(A - LC)| \simeq \beta |\lambda(A - BK)|$$

with  $3 \leq \beta \leq 10$ .